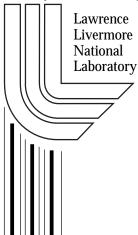
Fully Implicit Solution of Large-Scale Non-Equilibrium Radiation Diffusion with High Order Time Integration

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Fully Implicit Solution of Large-Scale

Non-Equilibrium Radiation Diffusion with

High Order Time Integration *

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Abstract

We present a solution method for fully implicit radiation diffusion problems dis-

cretized on meshes having millions of spatial zones. This solution method makes

use of high order in time integration techniques, inexact Newton-Krylov nonlinear

solvers, and multigrid preconditioners. We explore the advantages and disadvantages

of high order time integration methods for the fully implicit formulation on both

two- and three-dimensional problems with tabulated opacities and highly nonlinear

fusion source terms.

Key words: radiation diffusion, implicit methods, time integration, parallel

computing

PACS: 65M12, 65M20, 65Y05

1 Introduction

The simulation of nonlinear and coupled physical phenomena requires efficient numerical methods for solution. One application which is computationally intensive is modeling transport of neutral particles. Simulations of this application are important in calculations relevant to astrophysics, shielding, inertial confinement fusion, and atmospheric radiation. The Boltzmann transport equation is often used for modeling these problems. Due to the six degrees of freedom present in this equation, diffusion approximations are often used to give coarse estimates of solutions. Solution of the nonlinear diffusion approximation is still a demanding task, however, and the need for computational methods to efficiently solve these problems is still required.

In this paper, we present a fully implicit solution method for radiation diffusion problems. Fully implicit methods can allow for larger time steps and more accurate solves for a given amount of work than can explicit or semi-implicit methods. Our method makes use of high order in time integration techniques, inexact Newton–Krylov nonlinear solvers, and multigrid preconditioners. We target problems discretized on meshes having millions of spatial zones.

The first work on implicit solution of radiation diffusion problems was done by Axelrod, et.al. in 1984 when they presented a solution method using a high

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order GEAR ODE solver [1] for integration in time coupled with a Newton method and an approximate direct solver [2]. Their work showed a distinct advantage in both accuracy and run time of the implicit solver over an operator split approach on multigroup diffusion in one dimension. This work did not, however, follow with development of solvers for large-scale problems. In particular, the very effective Newton-Krylov methods [3–5] were not applied to implicit radiation diffusion problems until much later. One obstacle to use of these methods was the requirement of effective preconditioners for the linear Jacobian systems. In 1999 Rider, Knoll, and Olson applied these methods along with multigrid preconditioning to one- and two-dimensional equilibrium problems and showed substantial benefit in accuracy of a second order in time fully implicit solver over a second order in time semi-implicit solver where nonlinearities are lagged between time steps [6,7].

For semi-implicit methods, time steps are often chosen to be small enough to maintain accuracy despite explicit (or lagged) parts of the computation. When moving to an implicit method, however, the question of how to choose the time step should be revisited since an implicit method generally allows a larger step size for a given accuracy within a computation. Rider and Knoll suggested a method of choosing steps for the fully implicit formulation by using a hyperbolic model of the system [8]. Brown, et.al. simultaneously pursued implicit methods based on the work of Axelrod et.al. and employed ODE time integrator technology for three-dimensional multigroup diffusion [9]. These methods choose time steps based on solution accuracy requirements of the user and local time truncation error estimates [10].

Mousseau, Knoll, and Rider extended the work of Rider, Knoll, and Olson to problems where the radiation and matter fields are not in equilibrium.

They again applied Newton-Krylov methods with first and second order time stepping on two-dimensional problems [11,12]. They used an operator splitting method and multigrid for preconditioning. Brown and Woodward, also looking at non-equilibrium problems, developed an effective Schur complement-based preconditioner and demonstrated parallel scalability of a fully implicit solver based on high order time integration, Newton-Krylov methods, and semicoarsening multigrid techniques on three-dimensional problems with hundreds of millions of unknowns [13]. They found that preconditioners for the implicit system need to account for the coupling between matter and radiation effectively and also showed that the choice of preconditioner is crucial to the success of the fully implicit solve for large-scale problems.

More recently, implicit formulations of radiation diffusion have been combined with hydrodynamics problems in a coupled manner. Bates et.al. have developed a nonlinearly consistent solver for the coupled system [14]. Howell and Greenough have also extended implicit diffusion methods to include coupled problems with hydrodynamics within the context of adaptive mesh refinement [15]. Recent work in the area has also included investigations into the relative performance of Newton-Krylov methods with nonlinear multigrid techniques [16]. These studies have shown that Newton-Krylov is in general faster for these problems, but that nonlinear multigrid can have advantages in early time steps [17–19].

With the exception of the original work by Axelrod, et.al. and the work of Brown and Woodward, the above work does not consider high order time integration and tabulated opacities. High order methods can result in significant benefits in run time reduction while maintaining high accuracy. Care must be taken, however, in how these methods are applied as poor tolerance selection can lead to instabilities. In addition, when using Newton-Krylov methods, a nonlinear residual evaluation is required at each linear iteration. These evaluations require re-computation of opacity values from tables leading to possible discontinuities and computational expense. Lastly, none of the above work examines performance of implicit solvers for problems with highly nonlinear fusion source terms.

In this paper, we present a fully implicit solution method based on ODE time integration technology targeting large-scale radiation diffusion problems. We examine its advantages and disadvantages as compared to that of a semi-implicit method for three-dimensional problems with tabulated opacities from the LEOS equation of state library [20]. We also consider nonlinearities introduced from fusion source terms in the material energy equation. These terms are highly nonlinear and give rise to a potentially difficult nonlinear problem within the implicit formulation. Our results indicate that a fully implicit solution approach can achieve more accurate solutions than semi-implicit solution methods in many simulations involving the interaction of radiation and matter with highly nonlinear source terms. Furthermore, the fully implicit approach can be as cost effective as semi-implicit approaches in many cases despite the use of tabulated values for the opacities. Lastly, the solution approach is shown to scale well to very large problems solved on parallel machines.

In the next section of this paper, we outline the model problem we are considering. The section following overviews both our fully implicit method and a semi-implicit solution method to which we compare, as well as the nonlinear and linear solvers we use. In the results section, we give comparisons of the two methods, examinations of some algorithmic elements of our solution strategy, and some demonstrations of our method's performance on a large,

parallel machine. We conclude with some remarks about the viability of fully implicit methods on radiation diffusion problems.

2 Flux-Limited Radiation Diffusion Model

For this work, we consider the flux-limited, two-temperature formulation of radiation diffusion given by [21,22]

$$\frac{\partial E_R}{\partial t} = \nabla \cdot \left(\frac{c}{3\rho \kappa_R(T_R) + \frac{\|\nabla E_R\|}{E_R}} \nabla E_R \right) + c\rho \kappa_P(T_M) \cdot \left(aT_M^4 - E_R \right) + \chi(\mathbf{x}) caT_{\text{source}}^4, \tag{1}$$

where $E_R(\mathbf{x},t)$ is the radiation energy density $(\mathbf{x}=(x,y,z))$, $T_M(\mathbf{x},t)$ is the material temperature, $\rho(\mathbf{x})$ is the material density, c is the speed of light, and $a=4\sigma/c$ where σ is the Stephan–Boltzmann constant. The Rosseland opacity, κ_R , is a nonlinear function of the radiation temperature, T_R , which is defined by the relation $E_R=aT_R^4$. The Planck opacity, κ_P , is a nonlinear function of material temperature, T_M , which is related to the material energy through an equation of state, $E_M=\mathrm{EOS}(T_M)$. Here, T_{source} is a given source temperature, and $\chi(\mathbf{x})$ is a function of the spatial variable \mathbf{x} . In the limiter, the norm $\|\cdot\|$ is taken to be the l^2 norm of the gradient vector.

This equation is coupled to an equation expressing conservation of material energy given by

$$\frac{\partial E_M}{\partial t} = -c\rho\kappa_P(T_M) \cdot \left(aT_M^4 - E_R\right) + \mu(\mathbf{x}, t)T_M^5,\tag{2}$$

where μT_M^5 is a fusion source term with $\mu(\mathbf{x}, t)$ a function of both space and time.

This system is highly nonlinear due to the opacity dependences on temperatures as well as the fusion source term. Opacities typically depend on temperatures as a power law with typical expressions like $\kappa = CT^{-p}$ where C is a constant, and p may be 3-5 depending on the material and physical regime [23,24]. We consider Dirichlet, Neumann, and Robin boundary conditions for the system (1)–(2), and our focus here is on the development of solution methods for this system.

3 Solution Methods

For both the fully implicit and semi-implicit formulations, we employ a cellcentered finite difference approach for the spatial discretization. We use a tensor product grid with N_x, N_y , and N_z cells in the x, y, and z directions, respectively. Defining $E_{R,i,j,k}(t) \approx E_R(\mathbf{x}_{i,j,k},t)$ and $E_{M,i,j,k}(t) \approx E_M(\mathbf{x}_{i,j,k},t)$, with $\mathbf{x}_{i,j,k} = (x_i, y_j, z_k)$, and

$$\mathbf{E}_{R} \equiv \begin{pmatrix} E_{R,1,1,1} \\ \vdots \\ E_{R,N_{x},N_{y},N_{z}} \end{pmatrix} \text{ and } \mathbf{E}_{M} \equiv \begin{pmatrix} E_{M,1,1,1} \\ \vdots \\ E_{M,N_{x},N_{y},N_{z}} \end{pmatrix},$$

we can write our discrete equations in terms of a discrete diffusion operator given by

$$\mathbf{L}(\mathbf{E}_R) \equiv \left(L_{1,1,1}(\mathbf{E}_R), \cdots, L_{N_x, N_y, N_z}(\mathbf{E}_R)\right)^T, \tag{3}$$

a local coupling operator given by

$$\mathbf{S}(\mathbf{E}_R, \mathbf{E}_M) \equiv (S_{1,1,1}(\mathbf{E}_R, \mathbf{E}_M), \cdots, S_{N_x, N_y, N_z}(\mathbf{E}_R, \mathbf{E}_M))^T, \tag{4}$$

and a material source term

$$\mathbf{R}(\mathbf{E}_M) \equiv (R_{1,1,1}(\mathbf{E}_M), \cdots, R_{N_x, N_u, N_z}(\mathbf{E}_M))^T, \tag{5}$$

where

$$L_{i,j,k}(\mathbf{E}_{R}) \equiv \left(D_{i+1/2,j,k} \frac{E_{R,i+1,j,k} - E_{R,i,j,k}}{\Delta x_{i+1/2,j,k}} - D_{i-1/2,j,k} \frac{E_{R,i,j,k} - E_{R,i-1,j,k}}{\Delta x_{i-1/2,j,k}}\right) / \Delta x_{i} \quad (6)$$

$$+ \left(D_{i,j+1/2,k} \frac{E_{R,i,j+1,k} - E_{R,i,j,k}}{\Delta y_{i,j+1/2,k}} - D_{i,j-1/2,k} \frac{E_{R,i,j,k} - E_{R,i,j-1,k}}{\Delta y_{i,j-1/2,k}}\right) / \Delta y_{j}$$

$$+ \left(D_{i,j,k+1/2} \frac{E_{R,i,j,k+1} - E_{R,i,j,k}}{\Delta y_{i,j,k+1/2}} - D_{i,j,k-1/2} \frac{E_{R,i,j,k} - E_{R,i,j,k-1}}{\Delta z_{i,j,k-1/2}}\right) / \Delta z_{k}$$

with

$$\begin{split} D_{i+1/2,j,k} &\equiv \frac{c}{3\rho_{i+1/2,j,k}\kappa_{R,i+1/2,j,k} + \|\nabla E_R\|_{i+1/2,j,k}/E_{R,i+1/2,j,k}}, \\ D_{i-1/2,j,k} &\equiv \frac{c}{3\rho_{i-1/2,j,k}\kappa_{R,i-1/2,j,k} + \|\nabla E_R\|_{i-1/2,j,k}/E_{R,i-1/2,j,k}}, \\ D_{i,j+1/2,k} &\equiv \frac{c}{3\rho_{i,j+1/2,k}\kappa_{R,i,j+1/2,k} + \|\nabla E_R\|_{i,j+1/2,k}/E_{R,i,j+1/2,k}}, \\ D_{i,j-1/2,k} &\equiv \frac{c}{3\rho_{i,j-1/2,k}\kappa_{R,i,j+1/2,k} + \|\nabla E_R\|_{i,j-1/2,k}/E_{R,i,j-1/2,k}}, \\ D_{i,j,k+1/2} &\equiv \frac{c}{3\rho_{i,j,k+1/2}\kappa_{R,i,j,k+1/2} + \|\nabla E_R\|_{i,j,k+1/2}/E_{R,i,j,k+1/2}}, \\ D_{i,j,k-1/2} &\equiv \frac{c}{3\rho_{i,j,k-1/2}\kappa_{R,i,j,k-1/2} + \|\nabla E_R\|_{i,j,k-1/2}/E_{R,i,j,k-1/2}}, \end{split}$$

and

$$S_{i,j,k}(E_{R,i,j,k}, E_{M,i,j,k}) = c\rho_{i,j,k}\kappa_{P,i,j,k} \left(aT_{M,i,j,k}^4 - E_{R,i,j,k}\right), \text{ and}$$
 (7)

$$R_{i,j,k}(E_{M,i,j,k}) = \mu_{i,j,k} T_{M,i,j,k}^5.$$
(8)

Thus, our discrete scheme is to find $\mathbf{E}_R(t)$ and $\mathbf{E}_M(t)$ such that,

$$\frac{d\mathbf{E}_R}{dt} = \mathbf{L}(\mathbf{E}_R) + \mathbf{S}(\mathbf{E}_R, \mathbf{E}_M) + \mathbf{Q},\tag{9}$$

$$\frac{d\mathbf{E}_M}{dt} = -\mathbf{S}(\mathbf{E}_R, \mathbf{E}_M) + \mathbf{R}(\mathbf{E}_M),\tag{10}$$

where \mathbf{Q} includes the source term along with terms from the discretized boundary conditions. For more details, see [13].

3.1 Fully Implicit

For the fully implicit formulation, we use an ODE time integrator to handle the implicit time step selection for the system (9)-(10). In particular, we employ the parallel ODE solver, CVODE [10], developed at Lawrence Livermore National Laboratory and based on the VODPK package [25]. CVODE employs the fixed leading coefficient variant of the Backward Differentiation Formula (BDF) method [26,27] and allows for variation in the order of the time discretization as well as in the time step size.

The methods in CVODE are Predictor-Corrector in nature, and so each time step begins with the calculation of an explicit predictor. An implicit corrector is then employed to solve for the time step solution. This time integration technique leads to a coupled, nonlinear system of equations that must be solved at each time step. For example, solving the ODE system

$$\dot{y} = f(t, y), \tag{11}$$

with the backward Euler method (i.e., the BDF method of order 1), leads to

the following nonlinear system

$$0 = F(y) \equiv y - \Delta t f(t_n, y) - y_{n-1} \quad \left(i.e., \frac{y_n - y_{n-1}}{\Delta t} = f(t_n, y_n)\right)$$
(12)

that must be solved for $y = y_n$ at each time step. For the solution of this system, we use an inexact Newton–Krylov method with Jacobian-vector products approximated by finite differences of the form

$$F'(y)v \approx \frac{F(y+\theta v) - F(y)}{\theta},$$
 (13)

where θ is a scalar. Within the Newton-Krylov paradigm, only the implementation of the nonlinear function is necessary, and Jacobian matrix entries need never be formed or stored. Heuristic arguments for the case of systems arising from the implicit integration of ODEs show that $\theta = 1$ works quite well [28] and is the choice used in CVODE. Finally, the explicit predictor, $y_{n(0)}$, is used as an initial guess to the nonlinear system (12).

In the methods discussed above, we use the scaling technique incorporated into CVODE. Thus, we include an absolute tolerance (ATOL) for each unknown and a relative tolerance (RTOL) applied to all unknowns. These tolerances are then used to form a weight that is applied to each solution component during the time step from t_{n-1} to t_n . This weight is given as

$$w_i = \text{RTOL} \cdot |y_{n-1}^i| + \text{ATOL}_i, \tag{14}$$

and then the weighted root mean square norm

$$||y||_{\text{WRMS}} = \left[N^{-1} \sum_{i=1}^{N} (y_i/w_i)^2\right]^{1/2}$$
 (15)

is applied on all error-like vectors within the solution process. This scaling

gives each solution component equal weight when measuring the size of errors in y. For our application, we supply two absolute tolerances, one to be used with the radiation energy unknowns and one to be used with the material energy unknowns.

Time step sizes are chosen in an attempt to maximize step sizes while controlling the local truncation error, and thus give a solution that obeys a userspecified accuracy bound. The local truncation error (LTE) can, in general, only be estimated, and so CVODE uses the estimate

$$LTE(\Delta t^n) \equiv C_q(y_n - y_{n(0)}), \tag{16}$$

where y_n is the final iterate in the Newton iteration and C_q is a constant that depends on the BDF method order q but is independent of the solution. If $\|\text{LTE}(\Delta t^n)\|_{\text{WRMS}} < 1$, then the time step is accepted. If this condition is violated, the step size is cut, and the solution is recomputed. New steps are chosen by estimating the local truncation error at the new step, $\Delta t'$, as

$$\|\text{LTE}(\Delta t')\|_{\text{WRMS}} \approx \left(\frac{\Delta t'}{\Delta t^n}\right)^{q+1} \|\text{LTE}(\Delta t)\|_{\text{WRMS}},$$
 (17)

where q is the current method order. The new step is chosen to give the largest time step still satisfying $\|\text{LTE}(\Delta t')\|_{\text{WRMS}} < 1$. CVODE also changes the BDF method order by comparing the local truncation errors for the BDF methods of order q-1 and q+1 when using order q, and then taking the order that allows the largest time step.

We use the GMRES Krylov iterative solver for solution of the linear Jacobian system at each Newton iteration [29]. The tolerance for the Newton iteration is taken to guarantee that iteration error introduced from the nonlinear solver

is smaller than the local truncation error. The default linear system tolerance in CVODE is taken to be the factor $\alpha=0.05$ times the nonlinear system tolerance. This factor can be optionally set in the CVODE solver, and for some of the problems discussed below we use a smaller value of α , as the default of 0.05 did not work for the larger RTOL values. The default maximum subspace dimension for GMRES in CVODE is 5, and we use this default in all of our tests.

Preconditioning is generally essential when using Krylov linear solvers. To describe our preconditioning strategy, we begin by considering the content and structure of the Jacobian matrix. In (11), set $y = (\mathbf{E}_R^T, \mathbf{E}_M^T)^T$, and then form f using the right-hand sides of (9)-(10). The Jacobian matrices used in the Newton method are of the general form $F'(y) = (I - \gamma J)$, where $J = \partial f/\partial y$ is the Jacobian of the nonlinear function f, and the parameter $\gamma \equiv \Delta t \beta$ with Δt the current time step value and β a coefficient depending on the order of the BDF method. Recalling the definitions of the discrete divergence and source operators, the block form of the Jacobian of f is

$$J = \begin{pmatrix} \partial \mathbf{L}/\partial \mathbf{E}_R + \partial \mathbf{S}/\partial \mathbf{E}_R & \partial \mathbf{S}/\partial \mathbf{E}_M \\ -\partial \mathbf{S}/\partial \mathbf{E}_R & -\partial \mathbf{S}/\partial \mathbf{E}_M + \partial \mathbf{R}/\partial \mathbf{E}_M \end{pmatrix}$$
$$= \begin{pmatrix} A+G & B \\ -G & -B+C \end{pmatrix},$$

where $A = \partial \mathbf{L}/\partial \mathbf{E}_R$, $G = \partial \mathbf{S}/\partial \mathbf{E}_R$, $B = \partial \mathbf{S}/\partial \mathbf{E}_M$, and $C = \partial \mathbf{R}/\partial \mathbf{E}_M$. We note that G, B and C are diagonal matrices.

On close inspection of the nonlinear diffusion operator $\mathbf{L}(\mathbf{E}_R)$, we can write

$$\mathbf{L}(\mathbf{E}_R) = \hat{\mathbf{L}}(\mathbf{E}_R)\mathbf{E}_R,\tag{18}$$

where $\hat{\mathbf{L}}$ is a nonlinear matrix-valued function of \mathbf{E}_R . In all of our preconditioning strategies, we neglect the nonlinearity in the diffusion term and use the approximation

$$A = \partial \mathbf{L}(\hat{\mathbf{E}}_R) / \partial \mathbf{E}_R \approx \hat{\mathbf{L}}(\hat{\mathbf{E}}_R) \equiv \tilde{A},$$

where $\partial \mathbf{L}(\hat{\mathbf{E}}_R)/\partial \mathbf{E}_R$ is the Jacobian of \mathbf{L} evaluated at a radiation energy, $\hat{\mathbf{E}}_R$. The size of the neglected term is related to the derivatives of the Rosseland opacity and the flux-limiter. Our motivation for neglecting this term arises from the fact that $-\tilde{A}$ is symmetric and positive definite, whereas -A is not. In addition, the derivative of the flux-limiter may lead to numerical errors if $\nabla \mathbf{E}_R$ approaches 0.

Our preconditioning strategy is to factor the matrix

$$\begin{pmatrix} P & Q \\ U & T \end{pmatrix} \equiv \begin{pmatrix} I - \gamma(\tilde{A} + G) & -\gamma B + \gamma C \\ & & & \\ \gamma G & & I + \gamma B \end{pmatrix} = M$$

into the following:

$$M_{\mathrm{Schur}} = egin{pmatrix} I & QT^{-1} \\ 0 & I \end{pmatrix} egin{pmatrix} P - QT^{-1}U & 0 \\ & & \\ 0 & T \end{pmatrix} egin{pmatrix} I & 0 \\ & & \\ T^{-1}U & I \end{pmatrix}.$$

Letting $S = P - QT^{-1}U$, we write the solution to $M_{\text{Schur}}x = b$ as

$$\begin{pmatrix} x_1 \\ x_2 \end{pmatrix} = \begin{pmatrix} S^{-1}(b_1 - QT^{-1}b_2) \\ T^{-1}(-Ux_1 + b_2) \end{pmatrix}.$$

If the Schur complement, S, is exactly inverted, there will be no error associated with this preconditioner for the non-flux-limited, constant opacity case. In addition, because B, C and hence T is diagonal, there is no penalty associated with inverting T for every iteration of a method that inverts S, as there would be if a material energy diffusion term were added to the equations. Also note that S is formed by modifying the diagonal of P. Hence, we can employ multigrid methods to invert this Schur complement.

The Rosseland opacity will exhibit large changes where material interfaces exist in the domain. The temperature dependence gives rise to large value changes as well. These changes imply that the problem can be very heterogeneous. As a result, to invert matrix blocks formed from the diffusion operator, we use a multigrid method designed to handle large changes in problem coefficients. In particular, we use one V-cycle of the SMG algorithm developed by Schaffer [30,31] as our multigrid solver. Other multigrid methods have been developed for highly heterogeneous problems. A comparison of SMG and another of these methods can be found in [32]. We use SMG here because it is highly robust and scales well. Details of the SMG method can be found in the cited references. More information about multigrid methods in general can be found in [33].

Since Jacobian approximations can be expensive to compute, in CVODE the preconditioner is not updated with every Newton iteration. Preconditioner updates occur only when the Newton iteration fails to converge, 20 time steps pass without an update, or when there is a significant change in the time step size and order of the ODE method.

In summary, the main advantage of the fully implicit method is that we have accurate error control in the time step selection process allowing step sizes to automatically adjust to the problem physics while maintaining accuracy. The main disadvantage of the method is that opacities must be calculated for every linear iteration, as a nonlinear function evaluation is required in the matrix-vector product approximation (13). In general, fully implicit methods require more sophisticated solvers than semi-implicit methods. The solution method presented above has been tested on very large, three-dimensional problems and has been shown to be parallel scalable up to almost 6,000 processors [13].

3.2 Semi-Implicit

In the semi-implicit method we compare against, a backward Euler time stepping technique is applied, wherein opacities, flux-limiters, and material sources are evaluated at the start of a new time step using the solution from the previous step, and the coupling term is linearized about the solution from the previous step. The problem is put in a residual formulation so that the single linear solve required at each time step gives the increment to the solution values from the previous step's solution.

Beginning with the discrete system (9)–(10) and using (18), we can write

$$\frac{\mathbf{E}_{R}^{n+1} - \mathbf{E}_{R}^{n}}{\Delta t} = \hat{\mathbf{L}}(\mathbf{E}_{R}^{n})\mathbf{E}_{R}^{n+1} + \mathbf{K}(\mathbf{T}_{M}^{n})(a(\mathbf{T}_{M}^{n+1})^{4} - \mathbf{E}_{R}^{n+1}) + \mathbf{Q}^{n+1},$$
(19)

$$\frac{\mathbf{E}_M^{n+1} - \mathbf{E}_M^n}{\Delta t} = -\mathbf{K}(\mathbf{T}_M^n)(a(\mathbf{T}_M^{n+1})^4 - \mathbf{E}_R^{n+1}) + \mathbf{R}(\mathbf{T}_M^{n+1}),\tag{20}$$

where $\mathbf{K}(\mathbf{T}_{M}^{n})$ is a diagonal matrix with entries given by $K_{i,j,k} \equiv c\rho\kappa_{P}(T_{M,i,j,k}^{n})$ and $E_{M,i,j,k}^{n+1} = \mathrm{EOS}(T_{M,i,j,k}^{n+1})$. Next, letting $\mathbf{T}_{M}^{n+1} = \mathbf{T}_{M}^{n} + \Delta \mathbf{T}_{M}^{n}$ we linearize to obtain

$$(\mathbf{T}_M^{n+1})^4 = (\mathbf{T}_M^n + \mathbf{\Delta} \mathbf{T}_M^n)^4 \approx (\mathbf{T}_M^n)^4 + 4(\mathbf{T}_M^n)^3 \mathbf{\Delta} \mathbf{T}_M^n.$$

Similarly, we linearize $E_M = EOS(T_M)$ to obtain

$$\mathbf{E}_{M}^{n+1} = \mathbf{EOS}(\mathbf{T}_{M}^{n} + \mathbf{\Delta}\mathbf{T}_{M}^{n}) pprox \mathbf{EOS}(\mathbf{T}_{M}^{n}) + rac{\partial \mathbf{EOS}}{\partial \mathbf{T}_{M}}(\mathbf{T}_{M}^{n})\mathbf{\Delta}\mathbf{T}_{M}^{n},$$

or

$$\mathbf{E}_{M}^{n+1} - \mathbf{E}_{M}^{n} pprox rac{\partial \mathbf{EOS}}{\partial \mathbf{T}_{M}} (\mathbf{T}_{M}^{n}) \mathbf{\Delta} \mathbf{T}_{M}^{n}.$$

Thus,

$$(\mathbf{T}_M^{n+1})^4 \approx (\mathbf{T}_M^n)^4 + 4(\mathbf{T}_M^n)^3 \left[\frac{\partial \mathbf{EOS}}{\partial \mathbf{T}_M} (\mathbf{T}_M^n) \right]^{-1} (\mathbf{E}_M^{n+1} - \mathbf{E}_M^n).$$

We apply a similar linearization to the fusion source term, $\mu(T^{n+1})^5$. Substituting this last relationship into (19)–(20), we have

$$\mathbf{E}_{R}^{n+1} - \mathbf{E}_{R}^{n} = \Delta t \hat{\mathbf{L}}(\mathbf{E}_{R}^{n}) \mathbf{E}_{R}^{n+1} + \Delta t \mathbf{Q}^{n+1} + \Delta t \mathbf{K}(\mathbf{T}_{M}^{n}) \cdot \left(a \left[(\mathbf{T}_{M}^{n})^{4} + 4(\mathbf{T}_{M}^{n})^{3} \left[\frac{\partial \mathbf{EOS}}{\partial \mathbf{T}_{M}} (\mathbf{T}_{M}^{n}) \right]^{-1} (\mathbf{E}_{M}^{n+1} - \mathbf{E}_{M}^{n}) \right] - \mathbf{E}_{R}^{n+1} \right),$$
(21)

and

$$\mathbf{E}_{M}^{n+1} - \mathbf{E}_{M}^{n} = \Delta t \mu \left[(\mathbf{T}_{M}^{n})^{5} + 5(\mathbf{T}_{M}^{n})^{4} \cdot \left[\frac{\partial \mathbf{EOS}}{\partial \mathbf{T}_{M}} (\mathbf{T}_{M}^{n}) \right]^{-1} (\mathbf{E}_{M}^{n+1} - \mathbf{E}_{M}^{n}) \right] - \Delta t \mathbf{K} (\mathbf{T}_{M}^{n}) \cdot$$
(22)

$$\left(a\left[(\mathbf{T}_M^n)^4 + 4(\mathbf{T}_M^n)^3\left[\frac{\partial \mathbf{EOS}}{\partial \mathbf{T}_M}(\mathbf{T}_M^n)\right]^{-1}(\mathbf{E}_M^{n+1} - \mathbf{E}_M^n)\right] - \mathbf{E}_R^{n+1}\right),$$

where we solve for the changes, $\Delta \mathbf{E}_R^n \equiv \mathbf{E}_R^{n+1} - \mathbf{E}_R^n$ and $\Delta \mathbf{E}_M^n \equiv \mathbf{E}_M^{n+1} - \mathbf{E}_M^n$, given the previous values of \mathbf{E}_R^n and \mathbf{E}_M^n .

We solve the linear system (21)–(22) using the same linear solver as described above: the GMRES Krylov iterative solver with a Schur complement factorization preconditioner. The same multigrid method is used to invert the Schur complement matrix as in the fully implicit case. The linear iteration is performed until the relative residual is bounded by an input tolerance times the norm of the right-hand side,

$$\|\mathbf{r}\|_{\text{WRMS}} \le \epsilon \|\mathbf{b}\|_{\text{WRMS}},$$
 (23)

where \mathbf{r} is the linear system residual, \mathbf{b} is the linear system right-hand side, and ϵ is an input parameter. The WRMS norm is calculated in the same way as that for CVODE, using RTOL and ATOL values chosen as in the CVODE case.

Time steps are chosen to try to restrict changes in radiation energy and material temperature within a step. For specified minimum values, E_{\min} and T_{\min} , and specified fractional variations allowed in a step, E_{frac} and T_{frac} , the new step is computed by first calculating a maximum variation for each variable,

$$v_{R} = \max_{i,j,k} \left(\frac{\Delta E_{R}^{n}}{0.5(E_{R}^{n-1} + E_{R}^{n}) + E_{\min}} \right), \text{ and}$$

$$v_{M} = \max_{i,j,k} \left(\frac{\Delta T_{M}^{n}}{0.5(T_{M}^{n-1} + T_{M}^{n}) + T_{\min}} \right).$$

Then, the new step is chosen as

$$\Delta t^{\text{new}} = \Delta t^{\text{old}} \cdot \min(E_{\text{frac}}/(v_R + \delta), T_{\text{frac}}/(v_M + \delta)), \tag{24}$$

where $\delta=10^{-7}$ limits the maximum change in the step size. Note that this selection process is similar to the error control for the fully implicit case. However, while the semi-implicit approach bounds the maximum change in solution components over a time step, the fully implicit approach is bounding the maximum local truncation error made on a step with no direct control on the solution components. Finally, if the linear iteration fails to converge, then the step is repeated with $\Delta t^{\rm new}=0.5\cdot\Delta t^{\rm old}$.

To see the similarity, consider the radiation energy case. The semi-implicit method selects the step so that

$$v_R/E_{\text{frac}} = \frac{\Delta E_R^n}{0.5(E_R^{n-1} + E_R^n)E_{\text{frac}} + E_{\text{min}}E_{\text{frac}}} < 1.$$
 (25)

Taking ATOL_R = $E_{min}E_{frac}$, RTOL = E_{frac} , and noting that $0.5(E_R^{n-1} + E_R^n)$ is an approximation to the current radiation energy, we see that the *ith* component of the variation that is being bounded in the semi-implicit case is just,

$$\frac{\Delta E_R^n}{\text{RTOL} \cdot E_R^n + \text{ATOL}}.$$
 (26)

The variation in local truncation error that is bounded in the fully implicit case is just this expression. But instead of looking at the variation over a time step in ΔE_R^n , we look at the variation between the predictor and the corrector. Since the error in each of these approximations can be bounded, the variation in the predictor and corrector gives us a concrete estimate on

the local truncation error.

4 Numerical Results

In this section we present numerical results of solving nonlinear diffusion problems with the high order fully implicit method. We compare the method with a semi-implicit scheme for both accuracy and computational speed. We also investigate some of the advantages of the high order integration method and look at which orders give the highest benefit. Lastly, we show results of solving very large-scale nonlinear diffusion problems in parallel.

In the following subsections, the numerical statistical counters and parameters are

RTOL = relative tolerance,

MO = maximum order allowed, implicit method only

NST = time steps,

NNI = nonlinear iterations,

NLI = linear iterations,

RT = run time in seconds,

FAC = specified fractional variation allowed in both energies

within a step, semi-implicit only.

In the following examples we will use step and bi-cubic functions to define the

spatial and temporal extent of the source functions $\chi(\mathbf{x})$ and $\mu(\mathbf{x}, t)$ in (1) and (2) respectively, given by

$$H(x,\epsilon) \equiv \begin{cases} 1, & \text{if } |x| \leq \epsilon; \\ 0, & \text{otherwise.} \end{cases}$$

$$B(x,\epsilon) \equiv \begin{cases} 2\left(\frac{\epsilon+x}{2\epsilon}\right)^2 \left(6 - 8\frac{\epsilon+x}{2\epsilon}\right), & \text{if } -\epsilon < x \leq 0; \\ 2\left(\frac{\epsilon-x}{2\epsilon}\right)^2 \left(6 - 8\frac{\epsilon-x}{2\epsilon}\right), & \text{if } 0 \leq x < \epsilon; \\ 0, & \text{otherwise.} \end{cases}$$

4.1 Demonstration of accuracy

In this section, we present a test case which demonstrates accuracy of the fully implicit and semi-implicit codes compared to analytic solutions. All runs in this and the next subsection were done on a single processor in a Compaq cluster of 1 GHz EV68 Alpha processors.

Our first numerical test case is the Su-Olson problem, a one-dimensional Marshak problem that has a published analytic solution [34]. The problem starts with a homogeneous initial condition for the radiation and material temperature. A Robin boundary condition of the form,

$$E(\mathbf{0},t) - \left(\frac{2}{3\kappa_R}\right) \frac{\partial E(\mathbf{0},t)}{\partial \mathbf{x}} = 1,$$
(27)

is applied at $\mathbf{x} = 0$, and a homogeneous Dirichlet condition is applied at $\mathbf{x} = \infty$. In practice, this right-hand boundary condition is applied at $\mathbf{x} = 20$.

The material specific heat is given by, $c_v = \frac{a}{\epsilon} T_M^3$ and the equation of state is, $E_M(T_M) = \rho c_v T_M$. The flux limiter is not used in this simulation. The Planck and Rosseland opacities are both set to a constant, $\kappa_P = \kappa_R = 1.0 \ cm^2/g$, and $\epsilon = 0.1$. Heat is applied to the left-hand boundary as a result of the above

boundary condition. As the temperature of the radiation field increases, energy is transferred to the material. Simulations were run to a time of $3.34 \times 10^{-5} \mu s$. At this time the wave front is still far enough away from the right boundary that the boundary condition does not effect the solution.

Table 1 gives a comparison of implicit and semi-implicit method statistics for this problem. The maximum order, MO, was set to 5 for all implicit runs, and the relative errors reported are the maximum over the spatial grid computed at the end of the simulation. Relative error is with respect to analytic evaluations as described in Su and Olson [34].

The table shows that for each spatial grid, both methods produce approximately the same errors and that these errors are converging with the same rate as the grid spacing is refined. We also see that the discretization errors are independent of the time integration tolerances, indicating that the integration error is not polluting the spatial discretization error. Thus, the two codes have similar spatial discretization accuracies and we can consider that differences in solutions between the codes are related only to handling of time discretizations and nonlinear couplings.

4.2 Comparisons of fully and semi-implicit

In this section we present results of two 3D simulations with radiation sources, one in hydrogen and the other in carbon, and results of a 2D problem in hydrogen which includes a time-dependent material energy source. The 3D hydrogen problem is characterized by a rapid diffusion of radiation energy which will be

Table 1 Statistics for both fully implicit and semi-implicit solutions of the Su-Olson problem. Maximum relative error given is for radiation temperature.

Method	RTOL	FAC	NST	NNI	NLI	Max. Err.	RT		
200 grid points									
Implicit	10^{-5}	NA	1,609	1,654	1,604	4.43×10^{-2}	7.49		
Implicit	10^{-6}	NA	2,341	2,405	2,338	4.43×10^{-2}	9.54		
Semi-Imp.	NA	10^{-2}	3,919	NA	3,919	4.45×10^{-2}	15.5		
Semi-Imp.	NA	10^{-3}	39,053	NA	39,053	4.43×10^{-2}	72.9		
1,000 grid points									
Implicit	10^{-5}	NA	2,561	2,619	2,557	9.69×10^{-3}	84.6		
Implicit	10^{-6}	NA	3,759	3,832	3,755	9.69×10^{-3}	83.1		
Semi-Imp.	NA	10^{-2}	5,573	NA	5,573	9.91×10^{-3}	86.2		
Semi-Imp.	NA	10^{-3}	55,782	NA	55,782	9.64×10^{-3}	2,281.8		
10,000 grid points									
Implicit	10^{-5}	NA	3,788	3,853	3,781	1.05×10^{-3}	809.1		
Implicit	10^{-6}	NA	5,600	5,685	5,595	1.05×10^{-3}	1,332.1		
Semi-Imp.	NA	10^{-2}	7,621	NA	7,621	1.34×10^{-3}	1,190.7		
Semi-Imp.	NA	10^{-3}	76,540	NA	76,540	1.04×10^{-3}	17,935.2		

limited by the flux limiter. Diffusion in the 3D carbon problem is slower, and the flux limiter is of less importance. The 2D problem is characterized by a very fast heating rate due to the nonlinear source term. These test problems demonstrate the benefit and accuracy of fully implicit over semi-implicit.

4.2.1 Radiation source problem

In the 3D simulation, energy is supplied to the radiation field by a source with a specified black body temperature. The radiation energy is involved in four physical processes, heating from the source, diffusion of energy out of the heated region, transfer of energy to the material, and interaction with boundaries. The material energy is involved in only one process, heating via transfer of energy from the radiation field.

In these simulations we use the LEOS equation-of-state database [20] and a $20 \times 20 \times 20$ grid with 0.01 cm on each side. Homogeneous Neumann conditions are used on all boundaries, and the initial radiation and material temperatures are 15 eV.

The source is spherical positioned in a corner with a sharp boundary of radius $0.004 \ cm$,

$$\chi(\mathbf{x}) = H\left(\sqrt{(x - 0.01)^2 + (y - 0.01)^2 + (z - 0.01)^2}, 0.004\right).$$

The source temperature, T_{source} , is 300 eV.

These simulations were run for a short time interval of $10^{-6} \mu s$. This final time was kept short as these tests were designed to compare the two methods on transient problems. Although the grid used in these examples is coarse, the previous test case shows that errors due to the spatial approximation are the

same in both codes and hence cancel when the solutions are subtracted from each other.

Figure 1 shows the Planck and Rosseland opacities over the temperature ranges included in these runs. We see that the material energy for hydrogen varies more readily with temperature than it does with carbon. Similarly, we see the opacity values decrease faster with temperature for hydrogen than for carbon. These differences result in a more difficult problem for hydrogen than for carbon.

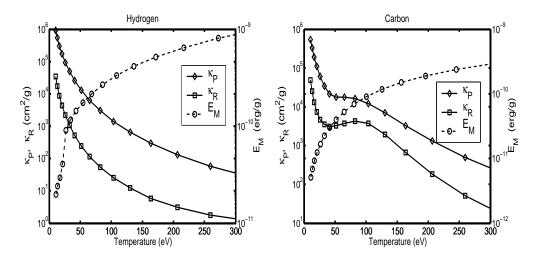


Fig. 1. Opacities over relevant temperature ranges in the 3D hydrogen (left) and carbon (right) simulations for $\rho = 1.0g/cc$.

Tables 2 and 3 summarize simulations using hydrogen and carbon, respectively. We see in all cases that allowing the implicit method to go to higher orders (above 2) results in a solution requiring fewer time steps than the second order scheme. Fewer steps are required because the integration method can take larger steps and lower the resulting error by using a higher order method. Comparing the implicit and semi-implicit methods in terms of computation run time and number of steps, the implicit method is faster than the semi-implicit for higher levels of requested accuracy. For the highly resolved

solutions, for example, implicit with $RTOL = 10^{-8}$ and semi-implicit with $FAC = 10^{-3}$, implicit can be several times faster.

For the hydrogen case, we see that the fully implicit method has trouble converging in a reasonable number of time steps for large values of RTOL. We believe this difficulty results from the method becoming numerically unstable. Because the larger tolerances allow more error in the solution, unstable modes can creep into the method and cause numerical instabilities. As RTOL is reduced, however, we see significant benefits to using the high order in time integration both in fewer numbers of time steps and also in decreased run time as compared to the second order method.

Figure 2 shows the relative error in the radiation temperatures for both the 3D hydrogen and carbon problems. In most cases, we see that going to higher order gives a more accurate solution than lower order for the fully implicit method. This accuracy difference results from the lower order method requiring smaller time steps in order to maintain accuracy. As a result, more steps are taken, and round off errors begin to accumulate. We also see that for all tolerances considered, the fully implicit method is more accurate than the semi-implicit method. Given that the semi-implicit run times are generally longer than that for the implicit method, significant speed benefits can be delivered with the implicit method.

Method	RTOL	МО	FAC	NST	NNI	NLI	RT
Implicit	10^{-5}	2	NA	DNF			
Implicit	10^{-5}	5	NA	DNF			
Implicit	10^{-6}	2	NA	1,673	1,728	4,798	370
Implicit	10^{-6}	5	NA	708	796	2,993	266
Implicit	10^{-7}	2	NA	3,681	3,784	8,900	726
Implicit	10^{-7}	5	NA	1,037	1,153	3,674	333
Implicit	10^{-8}	2	NA	7,924	8,166	17,085	2,089
Implicit	10^{-8}	5	NA	2,133	2,346	6,313	524
Semi-Imp.	NA	NA	10^{-1}	181	NA	1,013	100
Semi-Imp.	NA	NA	10^{-2}	1,807	NA	9,343	667
Semi-Imp.	NA	NA	10^{-3}	18,089	NA	78,166	6,595

4.2.2 Fusion source problem

Our next example is a 2D fusion source problem. In this problem we have added a material energy source which has a temperature dependence of T_M^5 . This is a good fit to a tritium-deuterium reaction rate at low temperature (less than a few keV) such as in a tokamak fusion experiment [35] (p. 29). The source function, $\mu(\mathbf{x}, t)$, is a product of a step function in cylindrical radius and a bi-cubic in time. The source, given in units of $\frac{erg}{cm^3} \frac{1}{s} \frac{1}{eV^5}$, which is positioned

Table 3
Statistics for 3D Carbon problem.

Method	RTOL	МО	FAC	NST	NNI	NLI	RT
Implicit	10^{-5}	2	NA	648	670	1,693	156
Implicit	10^{-5}	5	NA	401	412	1,099	88
Implicit	10^{-6}	2	NA	1,434	1,483	3,329	314
Implicit	10^{-6}	5	NA	807	852	2,024	192
Implicit	10^{-7}	2	NA	3,150	3,284	6,583	581
Implicit	10^{-7}	5	NA	1,658	1,758	3,969	377
Implicit	10^{-8}	2	NA	6,894	7,285	12,955	1,323
Implicit	10^{-8}	5	NA	3,462	3,832	8,112	985
Semi-Imp.	NA	NA	10^{-1}	149	NA	654	61
Semi-Imp.	NA	NA	10^{-2}	1,468	NA	4,986	561
Semi-Imp.	NA	NA	10^{-3}	14,653	NA	29,302	4,662

in the upper right corner of the domain with a sharp boundary in space is given by,

$$\mu(\mathbf{x},t) = \left(2.31 \times 10^{-11}\right) H\left(\sqrt{(x-0.01)^2 + (y-0.01)^2}, 0.005\right) \times B\left(t-10^{-8}, 10^{-8}\right).$$

These simulations in hydrogen use a 20×20 grid with 0.01 cm on each side. The initial temperatures for both radiation and material are $100 \ eV$, and the

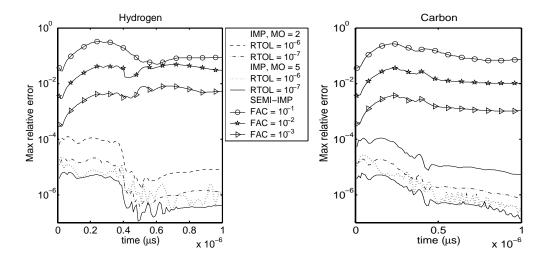


Fig. 2. Evolution of relative errors for radiation temperature in solution of 3D hydrogen (left) and carbon (right) problems. Relative error is with respect to the implicit simulation with $RTOL=10^{-8}$ and MO=5.

density was taken to be 1 gm/cm^3 . All boundary conditions are Neumann, and flux limiting is used for all runs. The LEOS equation of state package is used for opacity values. The simulations are run to a final time of 2.5×10^{-8} μs .

Figure 3 shows the history of the radiation and material source function at a point interior to the source region for this problem. For this problem energy is supplied to the material via the source term, transferred to the radiation and lost via diffusion. The strong nonlinear dependences of the source term on material temperature, T_M^5 , can lead to very rapid increases in temperature. In these problems the source is turned off by the time dependence in $\mu(\mathbf{x}, t)$.

The large heating rates can also lead to some problems with respect to automatic time step control. In the initial time period, before the source turns on, the automatic time step control for both methods will advance to the maximum allowed time step, HMAX. This advancement results in the system missing the turn-on of the source. In the simulations presented here, for both

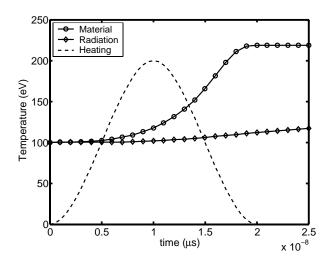


Fig. 3. Evolution of radiation and material temperatures as well as material source function (heating curve) profile in time for the 2D source problem.

implicit and semi-implicit methods, we bypass this problem by using a small maximum time step. It should be noted that in all runs shown in this section, this maximum step size only limits the step selection during the initial stages of the source. No limitations due to this parameter were observed for other times in the simulation. (A more consistent way of performing this simulation would be to have a third equation in the system which models the depletion of a fusion fuel density as its energy is added to the material. Solution of this three equation system will be the subject of a following paper.)

Tables 4 and 5 show results of our implicit and semi-implicit simulations for the material source problem, respectively. The values of RTOL, HMAX, and FAC have been chosen by trial-and-error to yield a set of runs with similar relative error. In the implicit runs we see that using higher order can lead to a reduction in run time by a factor of two or three for the more accurate small RTOL runs. For similar accuracy the semi-implicit is much slower. To some extent the poor performance of the semi-implicit method can be accounted for by the fact that this method required a much smaller value of HMAX to

Table 4
Statistics for implicit solution of 2D matter source problem.

RTOL	МО	HMAX	NST	NNI	NLI	RT
10^{-4}	2	10^{-9}	44	64	119	0.59
10^{-4}	5	10^{-9}	36	51	98	0.47
10^{-5}	2	10^{-9}	88	123	200	1.09
10^{-5}	5	10^{-9}	67	101	161	0.87
10^{-6}	2	10^{-9}	172	214	329	1.72
10^{-6}	5	10^{-9}	125	177	270	1.61
10^{-7}	2	10^{-9}	365	423	617	4.42
10^{-7}	5	10^{-9}	194	266	377	3.21
10^{-8}	2	10^{-9}	755	849	1212	9.14
10^{-8}	5	10^{-9}	300	403	538	3.00
10^{-9}	5	10^{-9}	513	669	852	5.01

resolve the time dependence of the source.

Figure 4 shows relative errors of the material temperatures for the 2D source problem. All methods show the same behavior, a significant increase in error once the source turns on, and a leveling off of the error once the source

Table 5
Statistics for semi-implicit solution of 2D matter source problem.

FAC	HMAX	NST	NLI	RT
10^{-3}	10^{-9}	904	1,635	17
10^{-4}	10^{-10}	9,754	17,388	191
10^{-5}	10^{-11}	96,884	130,481	1,539
10^{-6}	10^{-12}	751,153	751,152	11,744

turns off. All methods show convergence, with tolerance, to the highly resolved solution. Although not shown, relative errors in radiation temperature show similar results, but over a smaller scale.

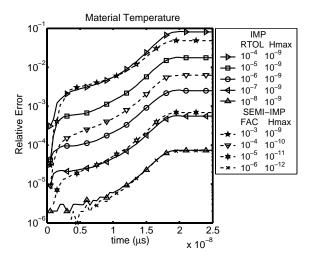


Fig. 4. Evolution of relative errors for material temperatures in the solution of the 2D source problem. Relative error is with respect to the implicit simulation with $RTOL = 10^{-9}$ and MO = 5 at a point interior to the source region.

4.3 Order Studies for the High Order Method

In this section, we explore the benefits of higher order and address the issue of which orders give the most gain in computational speed to solution for a given accuracy.

The first test is on a 3D fusion source problem. The source function, $\mu(\mathbf{x}, t)$, is a product of a bi-cubic function in spherical radius and a bi-cubic in time. The source positioned in the center of the domain and given in units of $\frac{erg}{cm^3} \frac{1}{s} \frac{1}{eV^5}$ is,

$$\begin{split} \mu(\mathbf{x},t) &= \left(4.75 \times 10^{-11}\right) B\left(t, 1.02 \times 10^{-8}\right) \times \\ &B\left(\sqrt{(x-0.005)^2 + (y-0.005)^2 + (z-0.005)^2}, 0.0025\right). \end{split}$$

These simulations in hydrogen use a $100 \times 100 \times 100$ grid with 0.01 cm on each side. The initial temperatures for both radiation and material were $100 \ eV$, and the density was taken to be $1 \ gm/cm^3$. All boundary conditions were Dirichlet with a $100 \ eV$ temperature, and flux limiting was used for all runs. The LEOS equation of state package was applied for opacity values, and the simulations were run to a final time of $2.5 \times 10^{-7} \ \mu s$. All runs were done on 8 processors of ASCI Frost.

Table 6 shows solver statistics for this problem with maximum order of the time integration set at 1, 2, 3, and 5. We see a significant gain in going from first to second order and in going from second to third order. Little benefit is seen in going to higher than third order. For this problem, third order is high enough to give the required accuracy and going higher introduces overhead in testing for changes to higher orders which will not give benefit.

Table 6
Statistics for implicit solution of 3D matter source problem.

RTOL	МО	NST	NNI	NLI	RT
10^{-7}	1	NA	NA	NA	DNF
10^{-7}	2	316	336	381	2,037
10^{-7}	3	168	187	187	1,310
10^{-7}	5	169	188	275	1,367
10^{-8}	1	5,193	5,408	5,395	29,218
10^{-8}	2	676	720	820	4,268
10^{-8}	3	360	408	549	2,736
10^{-8}	5	342	395	581	2,750

Figure 5 shows the solutions, heating time, and histories of time steps and order choices for a similar test problem with a density of $2 \ gm/cm^3$ and a final time of $2.5 \times 10^{-4} \ \mu s$. We used a higher density in this case to result in more transfer of energy from the matter to the radiation field. Also in this case we incerased the strength of the material heating source. We use the $\mu(\mathbf{x},t)$ of the above simulation with the leading constant factor increased to 9.5×10^{-11} . A longer time was used to give more information on order and step selection. In this problem, we clearly see the heating stage where the matter temperature undergoes significant increases then levels off while some of the energy is transferred to the radiation. Both field decrease in energy toward the end of the run as energy leaves the domain due to the Dirichlet boundary

condition.

The order history shows the method initially using third order as the heating region is traversed. After heating, the method reduces to second order for the majority of the rest of the run. The step size stays constant initially, then increases before the exponential portion of the heating begins. When the heating becomes significant, the step size is reduced, then increases again after the heating phase. The step continues to increase with a brief pause coinciding with a move to fourth order. These changes are a result of the method adjusting to the solution "settling down" after the large changes from heating and transfer of energy from matter to radiation. As the energy leaves the system, we see the order and step sizes change simultaneously. In general, a larger time step creates a larger error and a larger order reduces the error. Thus, we see that the method will raise the step and reduce the order, then adjust back. Toward the end of the run, we see these adjustments happen frequently as the solution shows energy decreases.

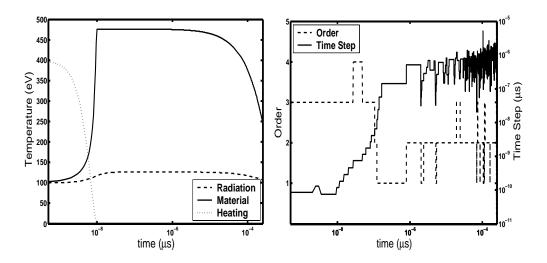


Fig. 5. Left: Heating time and temperature solutions for the 3D matter source problem. Temperatures are recoirded at the center of the source. Right: Time step and order history for implicit high order solution method.

These results show the ability of the variable order, variable step method to adjust to solution changes while maintaining a given requirement on the size of the local time integration truncation error. This type of adaptivity results in fewer time steps and thus lower round off error accumulation.

4.4 Results for Large-Scale Computations

In this section, we present results of parallel scalability studies for the fully implicit high order solution method. These studies give a measure of how well the solution method makes use of additional resources to solve larger problems. The first study was done with a constant opacity problem running on ASCI Red at Sandia National Laboratory. The next two studies were done using the LEOS tabulated opacity data base on radiation source and fusion source problems running on ASCI Frost at Lawrence Livermore National Laboratory.

In all studies in this section except the last, we used the PFMG multigrid method [36] rather than the SMG method mentioned earlier. We applied this algorithm because the PFMG method scales better to very large numbers of processors. Our last study was done with the SMG method and compares scaling results using this preconditioner with the PFMG method.

The first study ran on 5,832 processors of ASCI Red, an Intel machine with one processor per node running MPI for parallel communications. The system (1)-(2) was solved on the box $D \equiv \{(x, y, z) : 0 \le x, y, z \le 1cm\}$ with no matter source and a constant radiation source with $T_{source} = 300eV$ at the center of the domain. Constant Dirichlet conditions of T = 300K were applied on all boundaries. Initial conditions for both the radiation and matter were given by

Table 7
Statistics for scalability study on ASCI Red with constant opacity problem.

Processor	NST	NNI	NLI	RT	Avg. Cost	Step Scaled
Topology					per Step	Efficiency
$1 \times 1 \times 1$	123	140	186	2,485	20.2	100%
$2 \times 2 \times 2$	113	127	160	2,518	22.3	91%
$4 \times 4 \times 4$	105	119	154	2,424	23.1	88%
$8 \times 8 \times 8$	119	136	191	2,761	23.2	87%
$16 \times 16 \times 16$	116	129	212	2,970	25.6	79%
$18 \times 18 \times 18$	112	130	214	3,001	26.8	75%

 $T_{R,0} = T_{M,0} = 300K$. The equation of state was given simply as $T_M = E_M$, and the matter density was taken as $\rho = 1.0g/cc$. The Planck and Rosseland opacities were set constant and equal as $\kappa_P = \kappa_R = 10^5 cm^2/g$. Flux-limiting was turned on for all runs, and the simulation was run to 0.01s.

For this study, we added both unknowns and processors as we scaled up the problem keeping a spatial grid of $N_x = N_y = N_z = 40$ on each processor. Thus, problem size and computational resources were simultaneously increased.

Table 7 contains the results of the study. The reported scaled efficiency for a run on N processors was calculated by dividing the cost per step for the single processor run by the cost per step for the N processor case. As can be seen, all the statistics scaled extremely well. As this problem is dominated by the local coupling of the two fields (and not the diffusion operator), a high scaled efficiency is expected. In fact, we see 75% scaled efficiency for the largest test

case with 373.2M grid cells.

Our next scalability study was run on ASCI Frost with the LEOS equationof-state data base. The system (1)-(2) was solved on the box $D \equiv \{(x, y, z) : 0 \le x, y, z \le 0.01cm\}$ filled with carbon with no matter source and a constant radiation source with $T_{source} = 300eV$. The source which is positioned in the center of the domain with a sharp boundary and radius of 0.002 cm is given by,

$$\chi(\mathbf{x}) = H\left(\sqrt{(x - 0.005)^2 + (y - 0.005)^2 + (z - 0.005)^2}, 0.002\right).$$

Neumann condition are applied on all boundaries, and the initial temperature for radiation and material was 15 eV. No flux-limiting was applied. The problem was run to a final time of $2.5 \times 10^{-8} \ \mu s$ with a relative tolerance of 10^{-6} .

There were $40 \times 40 \times 40$ grid cells per processor, and we scaled up the number of processors from 1 to 448 giving a total of 28.67M grid cells. Since ASCI Frost has 16 processors per node and communication can be faster within a node than without, we used 8 processors per node for all but the two largest runs where we used 12 and 14 processors per node, respectively.

Table 8 shows the solver statistics and scaled efficiencies for this study. We see that as the problem size gets larger, the number of steps and solver iterations go up then decrease, but do not change dramatically. These results indicate that the solution method is able to solve these refined problems effectively. In addition, the run time does not significantly increase with scaling up the processors and unknowns simultaneously. We see a leveling off of the scaled efficiency for the total simulation run time at about 82%.

Table 8
Statistics for scalability study on ASCI Frost with tabulated opacity, radiation heating source problem.

Processor	NST	NNI	NLI	RT	Scaled	Scaled
Topology					Efficiency	Efficiency
					simulation	per step
$1 \times 1 \times 1$	489	521	901	1,401	100%	100%
$2 \times 1 \times 1$	563	590	1,098	1,673	84%	96%
$2 \times 2 \times 2$	588	611	1,085	1,718	82%	98%
$4 \times 2 \times 2$	559	579	1,177	1,836	76%	87%
$4 \times 4 \times 4$	529	548	1,082	1,716	82%	88%
$6 \times 6 \times 6$	479	498	983	1,735	81%	79%
$8 \times 8 \times 7$	463	482	958	1,704	82%	78%

Our last scalability study was run on ASCI Frost with the LEOS equation-of-state data base. The system (1)-(2) was solved on the box $D \equiv \{(x,y,z): 0 \le x,y,z \le 0.01cm\}$ with no radiation source and a nonlinear material heating source. The box was filled with hydrogen, and flux-limiting was applied. The material heating source, given in units of $\frac{erg}{cm^3}\frac{1}{s}\frac{1}{eV^5}$, is positioned in the center of the domain with a smooth boundary at a radius of 0.0025 cm and is given by

$$\mu(\mathbf{x},t) = \left(4.75 \times 10^{-11}\right) B\left(t, 10^{-8}\right) \times \\ B\left(\sqrt{(x - 0.005)^2 + (y - 0.005)^2 + (z - 0.005)^2}, 0.0025\right).$$

In this simulation, the source is turned off by the time dependence in $\mu(x,t)$. The time dependence of this function is a maximum at t=0 then turned off with a half width in time of $1.0 \times 10^{-8} \ \mu s$. Neumann conditions were applied on all boundaries. The initial temperature for radiation and material was $100 \ eV$. The problem was run to a final time of $2.5 \times 10^{-8} \ \mu s$ with a relative tolerance of 10^{-7} .

Similar to the previous example, we used $40 \times 40 \times 40$ grid cells per processor and scaled up the number of processors from 1 to 448 with the same numbers of processors per node in use.

We performed two scalability studies for this test problem. In the first, we applied the PFMG multigrid method to solve the Schur complement system (3.1). Results for this study are found in Table 9 where we see a scaled efficiency of the simulation of about 67% for 448 processors. Table 10 contains the results of the same study but with the SMG method applied to solve the Schur complement system. Here the scaled efficiencies are still decreasing and are at 46% for 448 processors. The difference between these two studies is due to the fact that the SMG solver includes more coupling in the method and thus requires more parallel communication. The increased couplings result in better algorithmic scaling, as can be seen from the nearly level numbers of time steps, nonlinear iterations, and linear iterations with the SMG preconditioner, but results in less parallel efficiency. These two scaling studies show a classic tradeoff between algorithm and implementation scalabilities. We further note that both the PFMG and SMG preconditioners used in these studies were taken from the hypre library [37,38], and the developers of this library indicated that they have seen similar differences between the two methods.

Table 9
Statistics for scalability study on ASCI Frost with tabulated opacity, material heating source problem, PFMG multigrid method.

Processor	NST	NNI	NLI	RT	Scaled	Scaled
Topology					Efficiency	Efficiency
					simulation	per step
$1 \times 1 \times 1$	124	145	258	439	100%	100%
$2\times1\times1$	118	136	278	454	97%	92%
$2 \times 2 \times 2$	112	129	255	431	102%	92%
$4 \times 2 \times 2$	133	142	333	529	93%	89%
$4 \times 4 \times 4$	120	134	259	459	96%	93%
$6 \times 6 \times 6$	124	138	269	500	88%	88%
$8 \times 8 \times 7$	144	156	338	654	67%	78%

5 Conclusions

We have presented a fully implicit solution method for radiation diffusion problems with highly nonlinear sources. Our method makes use of high order in time integration techniques, inexact Newton-Krylov nonlinear solvers, and multigrid preconditioning. We have incorporated the use of tabular opacities in our model in an effort to enhance the accuracy of our test problems as well as to evaluate the added costs of additional function evaluations in the fully implicit approach. Our results indicate that a fully implicit solution approach can achieve more accurate solutions than semi-implicit solution methods in

Table 10
Statistics for scalability study on ASCI Frost with tabulated opacity, material heating source problem, SMG multigrid method.

Processor	NST	NNI	NLI	RT	Scaled	Scaled
Topology					Efficiency	Efficiency
					simulation	per step
$1 \times 1 \times 1$	117	141	164	393	100%	100%
$2 \times 1 \times 1$	110	127	171	419	94%	88%
$2 \times 2 \times 2$	101	115	150	414	95%	82%
$4 \times 2 \times 2$	99	110	151	448	88%	74%
$4 \times 4 \times 4$	97	109	156	531	74%	61%
$6 \times 6 \times 6$	106	119	165	688	57%	52%
$8 \times 8 \times 7$	101	113	163	858	46%	40%

many simulations involving the interaction of radiation and matter with highly nonlinear source terms. Furthermore, the fully implicit approach can be as cost effective as semi-implicit approaches in many cases despite the use of tabulated values for the opacities. Lastly, the solution approach is shown to scale well to very large problems solved on parallel machines.

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